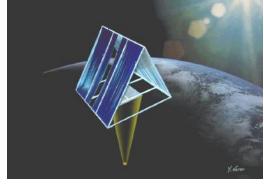
Engineering Research for New Tethered Solar Power Satellite

- New Concept of Tethered-SPS
- Engineering Aspects
 - 1 Attitude Stabilization
 - 2 Modularization
 - 3 Thermal Characteristics
 - 4 Construction and Maintenance
 - **5 Evolutionary Development**
 - **6 Easy Investment**
 - 7 Coexistence with Other Geostationary Satellites

Typical Examples of SPS



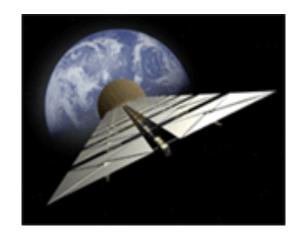
NASA Reference System



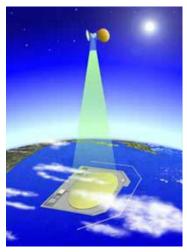
ISAS SPS 2000



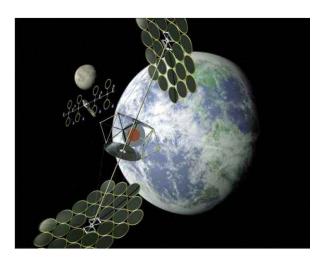
NASA Sun Tower



European Solar Sail Tower



NASDA (JAXA) SSPS Model



NASA Integrated Symmetrical Concentrator

Technical difficulties/feasibility problems in the past models

Most difficult point: to direct large solar panel to the sun while transmitting antenna be pointed to rectenna on ground

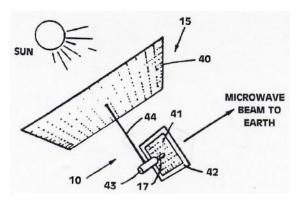
⇒ movable mechanism, rotary joint or rotating mechanism for mirrors, are required.

However,

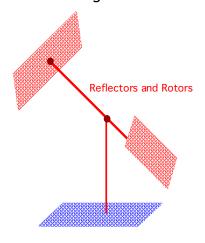
Rotary joint: no practical technologies without serious power loss

Rotating mirror: very difficult technologies for attitude control and stabilization of the large-thin film structure

Movable mechanism: one-point failure problem



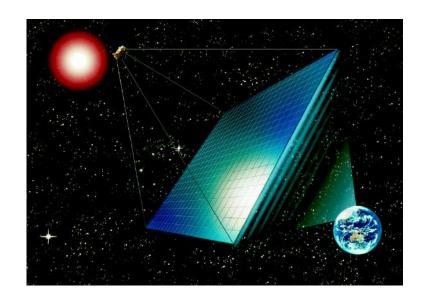
Glaser's original idea



Sun Pointing System

Why Tethered-SPS?

- ⇒ cost analysis is not possible if technical feasibility is not assured
- ⇒ need to alter our perspective on the SPS system configuration: simple and feasible configuration
 - ⇒ Tethered-SPS
 Automatically stabilized by gravity gradient force.
 No sun-pointing mechanism.
 Less power efficiency but robust and costless.

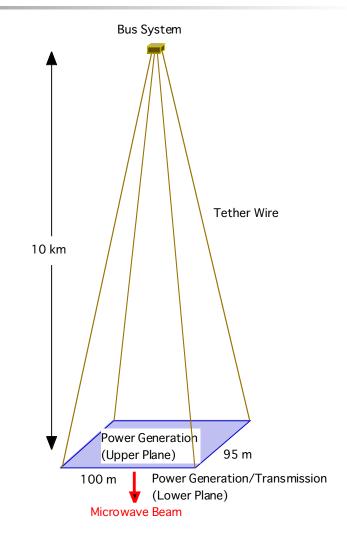


Unit of Tethered-SPS

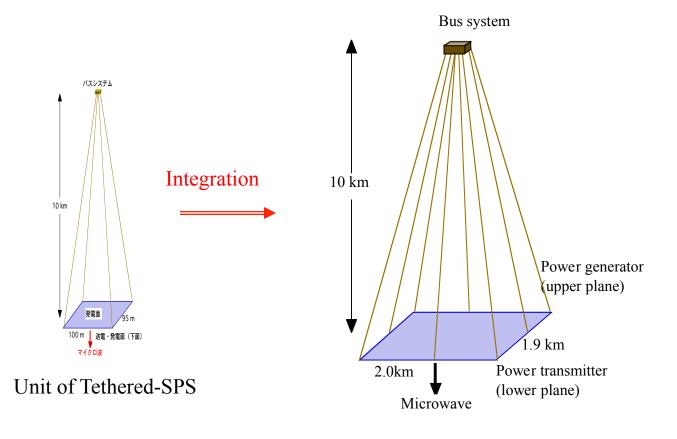
Total Weight: 42.5 MT

Tether Length: 2-10 km

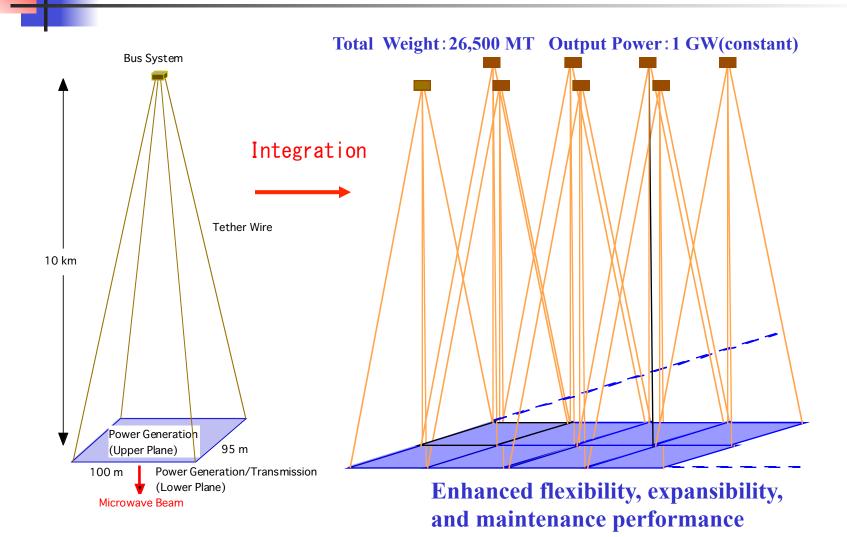
Output Power: 2.2 MW



Concept of Tethered-SPS (Former Type)



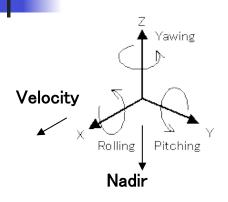
Concept of Tethered-SPS (New Type)



System Characteristics of Tethered SPS (Constant Power)

	Item	Performance	Note
Tethered SPS	Weight	26,562 MT	25,234MT (Panel) , 1,328MT (Bus)
	Size	2.5 km x 2.375 km x 10 km	250 Tethered SPS modules
	Output Power	1.36 GW	Microwave frequency 5.8 GHz
Tethered SPS Module	Weight	42.5 MT	40,375 kg(Panel), 2,125 kg(Bus)
	Tether Length	2-10 km	Width 1cm, Para-aramid fiber (Kevlar/DuPont), UV protection coating
	Panel Size	100m x 95 m	100x 95 Power generation/ Transmission modules
	Output Power	2.2 MW	Microwave frequency 2.45 GHz
Power Generation/ Transmission Module	Weight	4.25 kg	Microwave circuit 2.3kg(10g/W) Solar cell 0.45kg(0.5g/W) Batteriies 1.0kg(2000Wh/kg) Structure 0.5kg(0.025g/cc)
	Size	1 m x 1m x 2 cm	
	Output Power	230 W	Microwave frequency 2.45 GHz





Amplitude of attitude fluctuation induced by solar radiation pressure when deviation between the center of gravity and the center of shape is 1 %.

Direction of Fluctuation	Amplitude [rad]
Pitch angle by disturbance along pitch axis	2.8x10 ⁻³ (0.16 degrees)
Roll angle by disturbance along roll axis	1.3x10 ⁻⁴
Yaw angle by disturbance along roll axis	1.2x10 ⁻⁴
Roll angle by disturbance along yaw axis	5.8x10 ⁻⁵
Yaw angle by disturbance along yaw axis	2.4x10 ⁻³



Balancing of gravity gradient force and centrifugal force

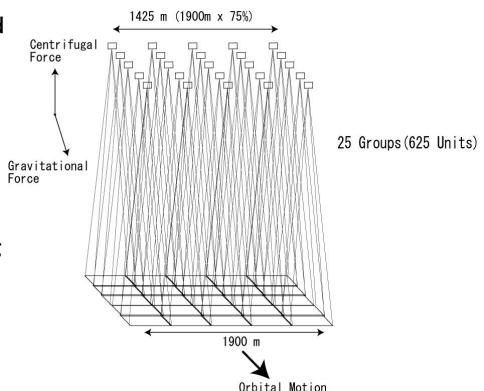
Panel size: P m (475 m)

Distance of bus: 0.75P m (356 m)

Allowable tether fluctuation:

 $0.75P/L (8.2^{\circ})$

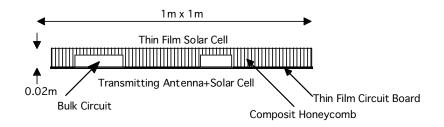
(): calculation for 5 x 5 unit grouping



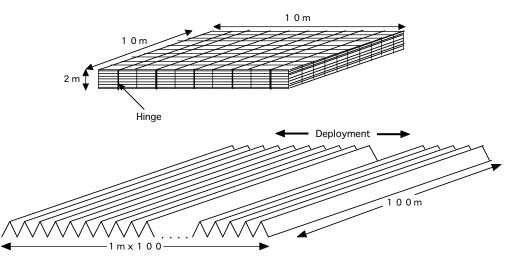


(2) Modularization

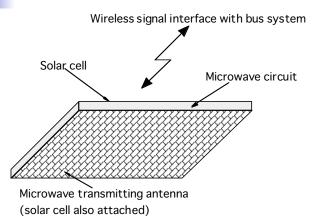
- Power generation and transmission panel is composed of perfectly equivalent modules.
- •Power generated by the solar cell is converted into microwave power in a module.
- •All modules are controlled by wireless LAN.
- No power/signal cabling between the modules.
- ⇒ Robust and low cost (mass production, easy quality control)



Cargo in transportation vehicle



Power Generation/Transmission Module



Power generation/transmission module



Functional model of module (solar cell side)



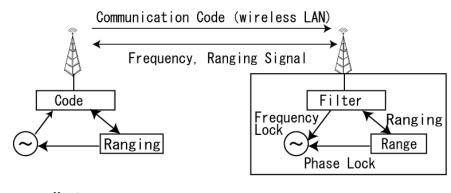
Mock-up of module



Functional model of module (microwave antenna side)



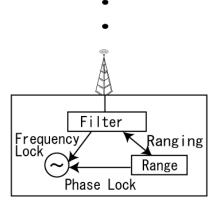
Synchronization of frequency and phase between separated bus units



Master

Equivalent Circuit for Master and Slave units.

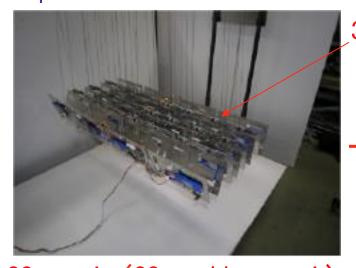
Mater unit sequentially synchronizes the frequency and phase of Slave unit with those of Master unit.



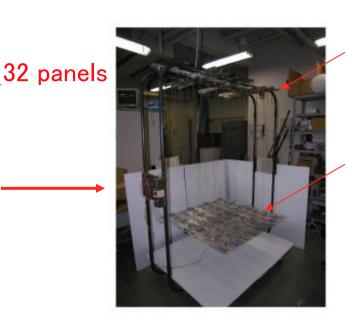
Slave

Concept of Simple Deployment Mechanism using SMA (Shape Memory Alloy) Actuators

Testing using miniature panel model



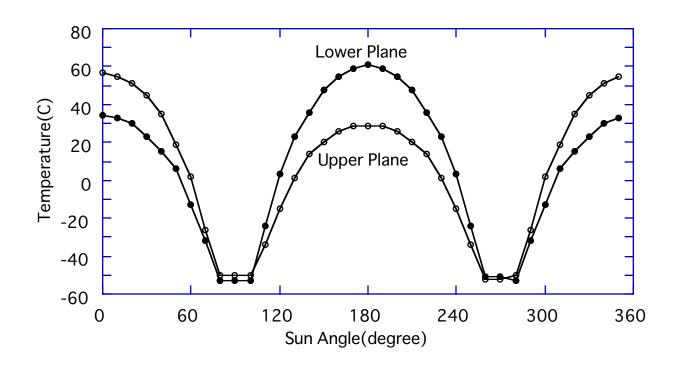
32 panels (23cmx11cm each) during deployment



Cancellation mechanism for gravity field

32 panels after deployment

(3) Temperature variation of the upper and lower planes





(4) Construction and Maintenance

LEO to GEO

Cargo

Transportation from Ground

transfer

OTV

to LEO

RLV

1. Unmanned construction

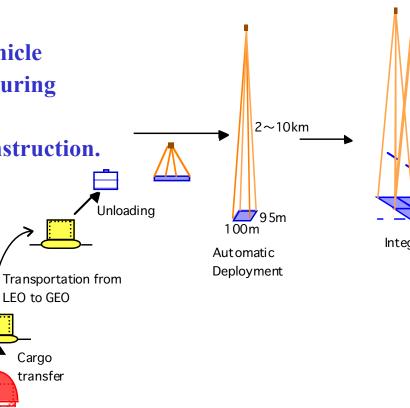
2. Moderate orbit transfer vehicle

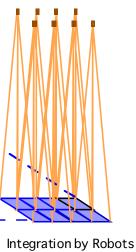
3. No radiation degradation during transportation

4. Full verification during construction.

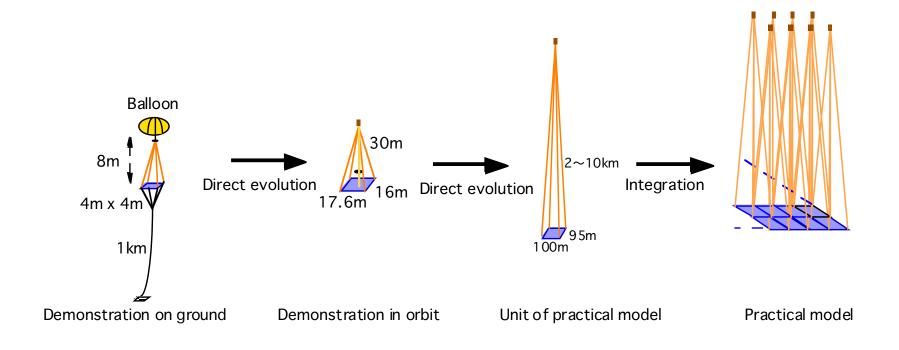
5. Easy maintenance and

scale-up

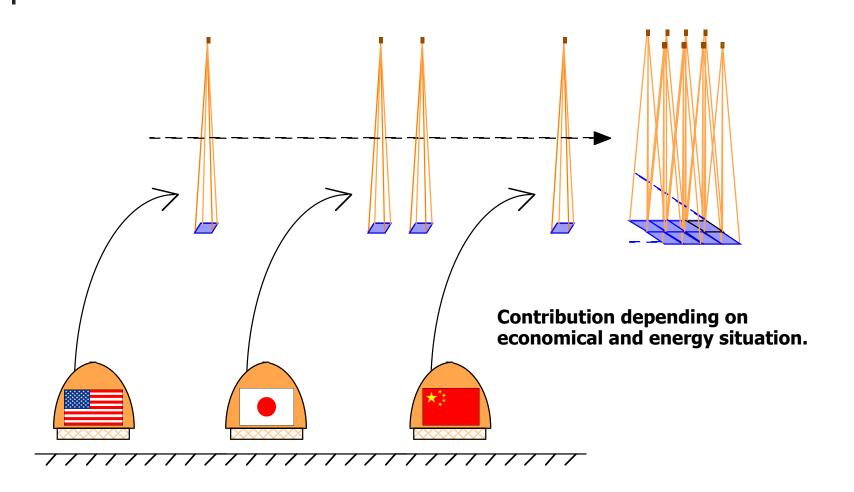




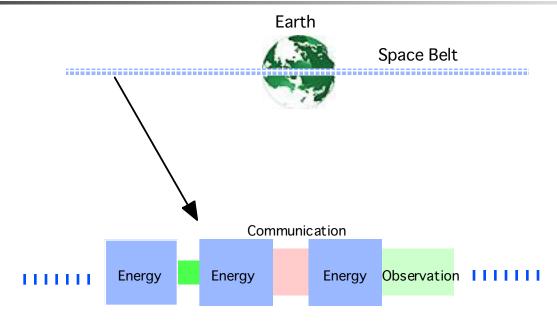
(5) Evolutionary Development from Demonstration Model to Commercial Model



(6) Easy Investment with Clear Work Interface



(7) Coexistence with Other Geo-stationary Satellites



Complex of facilities for Energy, Communication, Earth Observation, Space Telescope, Space Experiment, and Maintenance in Geosynchronous orbit.

World primary energy(13,000 GW) can be supplied from space belt with the length of 32,500 km (14 % of total space belt).

Technologies required for Tethered-SPS

Solar cel 1	Power efficiency 35 %, 2 kW/kg, 0.5 kW/m ² , ¥50/W		
Microwave	DC/RF conversion efficiency 85 %, 10 g/W, ¥100/W,		
transmissi o n	Beam control technologies for targeting at 3.5		
	km-diameter rectenna more than 90 % power efficiency		
Power storage	1.5 kWh/kg, ¥10/Wh, Charge/discharge efficiency 90 %,		
	DOD 50 %, Life 30,000 charge/discharg e		
Microwave reception	RF/DC conversion efficiency 85 %, ¥50/W		
Transportati o n	¥15,000/kg (ground to LEO, LEO to GEO)		

Summary and Conclusion

- New model Tethered-SPS is a highly practical SPS concept, with a number of advantages in the production, integration, construction, operation, and maintenance, as compared with the past SPS models.
- Since the technologies employed in the Tethered-SPS are essentially achievable, this model can be used as a realistic reference model to evaluate the cost and CO₂ load as a future energy system.
- •Further investigations are required to confirm the technical feasibilities, especially for microwave control, integration of the units using robotics, and orbit maintenance of the large structure.